

IN THE CLAIMS

Please amend the claims as follows:

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Claim 1 (Currently Amended) A blade, of a gas turbine, having a ~~wide turning angle~~, ~~said blade having~~ a belly side, a back side, a front edge, and a rear edge, the blade comprising a wide turning angle greater than 120 degrees, wherein ~~diameter~~ diameters of circles inscribing the belly side and the back side of adjacent blades decrease gradually from the front edge to the rear edge.

Claim 2 (Currently Amended) The blade according to claim 1, wherein ~~the turning angle is 115 degrees or more~~, a ratio of blade maximum wall thickness and blade chordal length is 0.15 or more, and a wedge angle of the rear edge is 10 degrees or less.

A' Claim 3 (Original) The blade according to claim 1, wherein the blade is a cooling blade of which cooling passage is near the rear edge, and the ratio of wall thickness of rear edge and throat between adjacent blades is 0.15 or less.

Claim 4 (Original) The blade according to claim 1, wherein said blade is a cooling blade of which cooling passage is near the rear edge, and the ratio of the distance from the cooling passage to the rear edge and the wall thickness of rear edge of the blade is 2 or less.

Claim 5 (New) A blade, of a gas turbine, having a belly side, a back side, a front edge, and a rear edge, wherein diameters of circles inscribing the belly side and the back side of adjacent blades decrease gradually from the front edge to the rear edge, a ratio of blade maximum wall thickness and blade chordal length is 0.15 or more, and a wedge angle of the rear edge is 10 degrees or less.

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